

# **NASA2 Benchmark**

## **Comparison of Analysis Results to Date**

*Case Manager*

William M. Humphreys, Jr.  
*Advanced Measurements and Data Systems Branch*

***Phased Array Methods Panel Session***  
***6<sup>th</sup> Berlin Beamforming Conference***  
***Berlin, Germany***

***29 February, 2016***



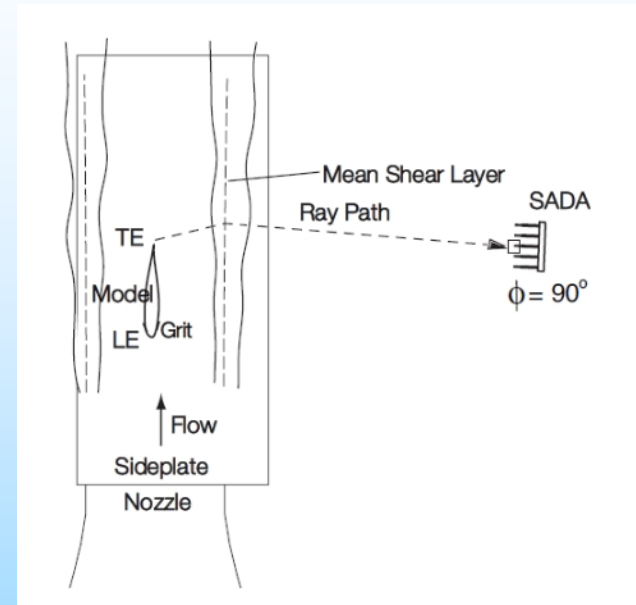
## ***NASA2 Benchmark***

- Leading Edge / Trailing Edge Noise Test
- Conducted in LaRC Quiet Flow Facility
- NACA 63-215 Airfoil:
  - 16-inch (40.6-cm) Chord
  - 36-inch (91.4-cm) Span
  - Angle of attack = -1.2 degrees
- Small Aperture Directional Array (SADA)
  - 33 B&K 4138 1/8" microphones
  - Positioned at 90 degrees wrt trailing edge of model



# NASA2 Benchmark

- Mach 0.17
- Total Temp = 79.5° F (26.4° C)
- DAS Sample Rate = 142857.71 Hz
- 2048000 Samples Acquired
- Background Runs Acquired
- Flat Mean Shear Layer
- Benchmark Updates for Year 2:
  - Addition of array shading coefficients to dataset
  - Addition of shear layer correction / steering vectors to dataset



## ***Analysis Methods Contributors***

- University of Adelaide, South Australia (2015)  
(POC: Ric Porteous)
- Laboratoire Vibrations Acoustique (LVA), INSA-Lyon, France (2016)  
(POC: Antonio Pereira)
- NASA Langley (2015, 2016)  
(POC: Christopher Bahr)



***Original DAMAS Analysis Results – 2004  
(Tom Brooks and William Humphreys, NASA LaRC)***

***Reference: Brooks and Humphreys, “A Deconvolution Approach for the Mapping of Acoustic Sources (DAMAS) Determined from Phased Microphone Arrays”, Journal of Sound and Vibration, Volume 294, 2006.***



# Original DAMAS Analysis Results – 2004

$$\hat{G}_{array} = \begin{bmatrix} G_{m1m1} & G_{m1m2} & \dots & G_{m1mM} \\ G_{m2m1} & \ddots & & \vdots \\ \vdots & & \ddots & \vdots \\ G_{mMm1} & \dots & \dots & G_{mMmM} \end{bmatrix} \rightarrow Y(\hat{e}) = \frac{\hat{e}^T \hat{W} (\hat{G}_{array})_{diag=0} \hat{W}^T \hat{e}}{(\sum_{m=1}^M w_m)^2 - \sum_{m=1}^M w_m^2}$$

Cross Spectral Matrix (CSM)

Conventional FD Beamformer  
with CSM Diagonal Removal

Desired Source Distribution      BF Source Distribution



$$\hat{X} = \hat{A}^{-1} \hat{Y}$$



Influence Matrix



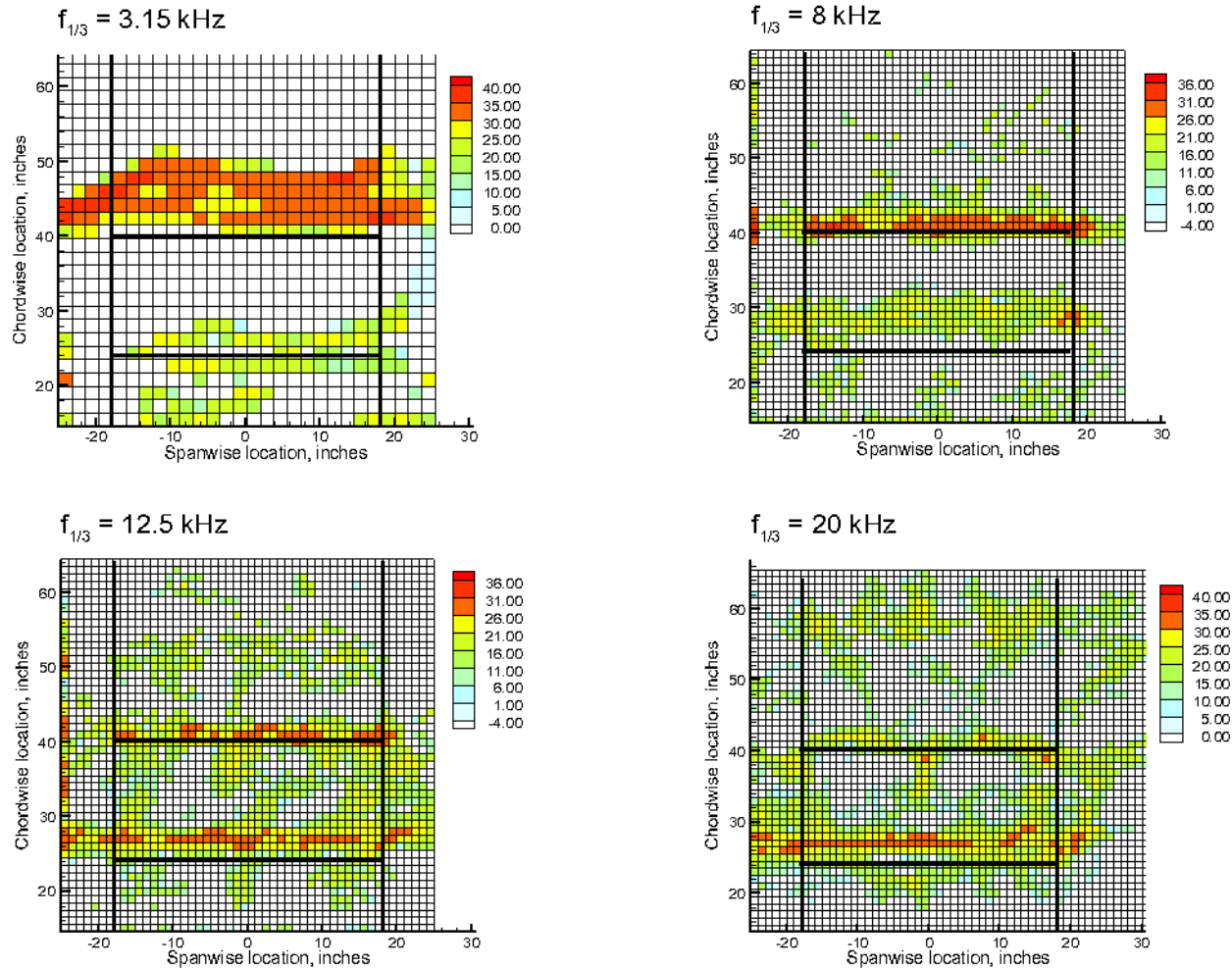
$$X_n^{(i)} = Y_n - \left[ \sum_{n'=1}^{n-1} A_{nn'} X_{n'}^{(i)} + \sum_{n'=n+1}^N A_{nn'} X_{n'}^{(i-1)} \right]$$

Gauss-Seidel Solver

- Shear layer correction – modified Amiet method (Humphreys et al. ,1998)
- Beamform Integrations – method of Brooks and Humphreys (1999)
- DAMAS Integrations - direct summation of deconvolved source maps

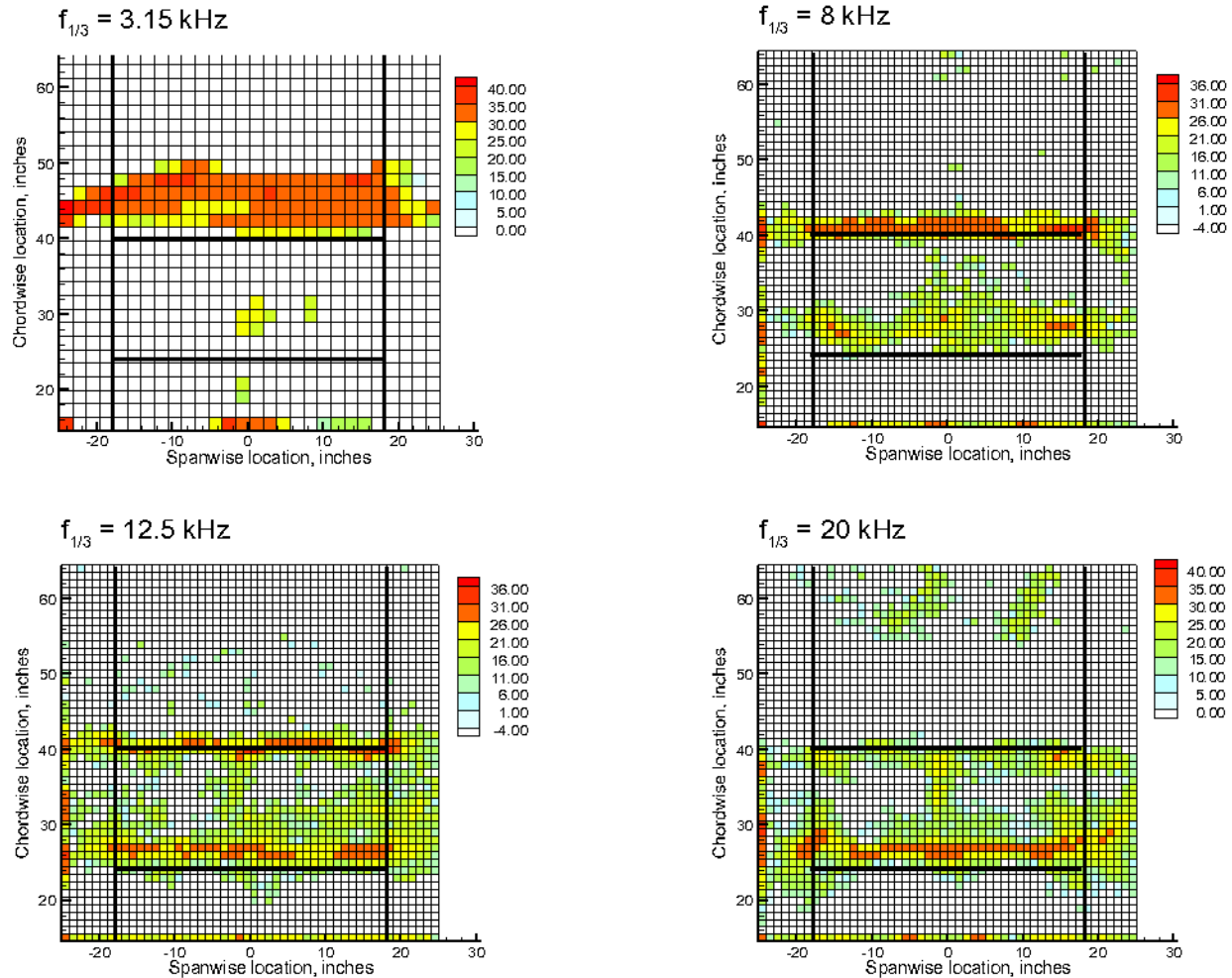
# Original DAMAS Analysis Results – 2004

Figure 11 - LE / TE CASE, T1390R15 - T1432R15 (no airfoil), Standard DAMAS



# Original DAMAS Analysis Results – 2004

Figure 13 - LE / TE CASE, T1390R15 - T1432R15 (no airfoil), DR DAMAS





# Original DAMAS Analysis Results – 2004

- Spatial Integrations

— SADA TE

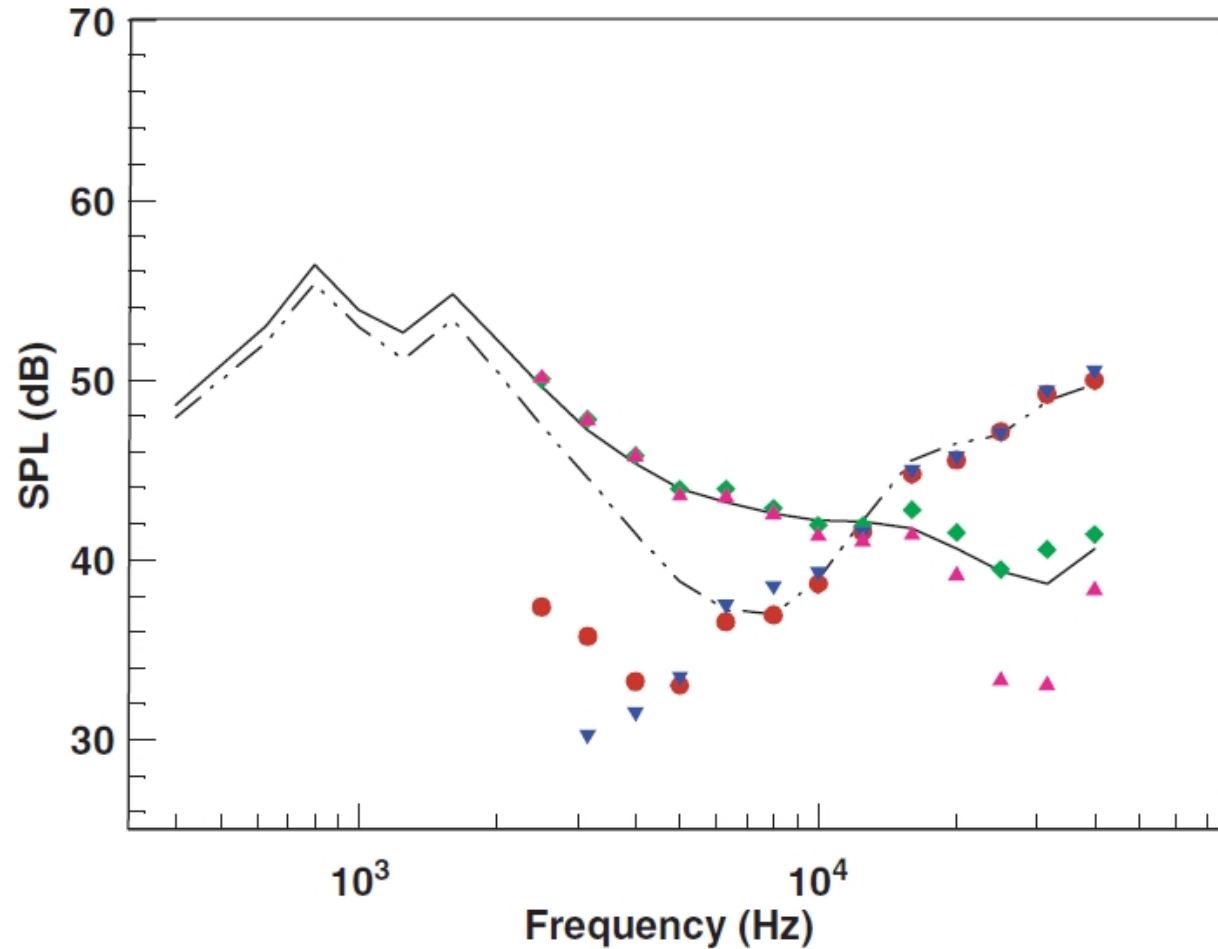
- · - · SADA LE

● STD DAMAS LE

◆ STD DAMAS TE

▼ DR DAMAS LE

▲ DR DAMAS TE



**Results from University of Adelaide /  
University of New South Wales  
(*Ric Porteous, Adelaide  
Zebb Prime, Con Doolan, and Danielle Moreau, UNSW*)**

Originally Presented at Dallas, 2015 Workshop



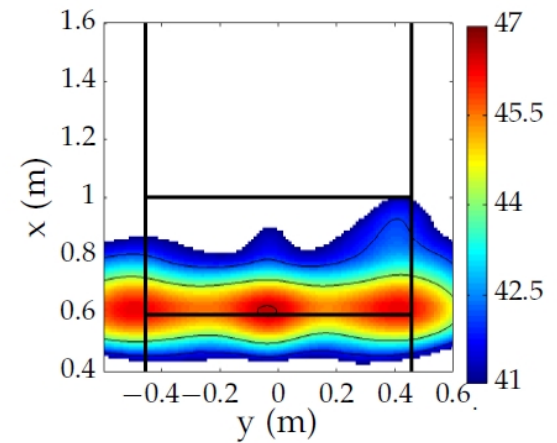
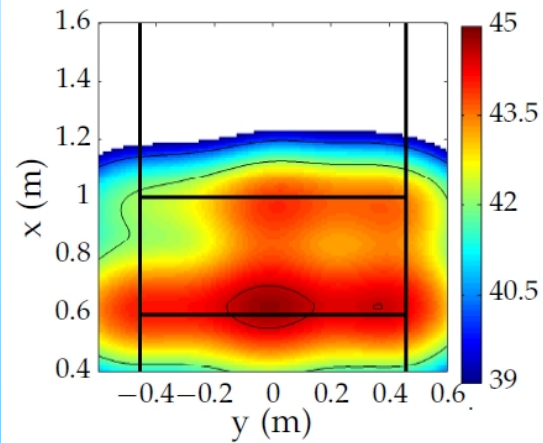
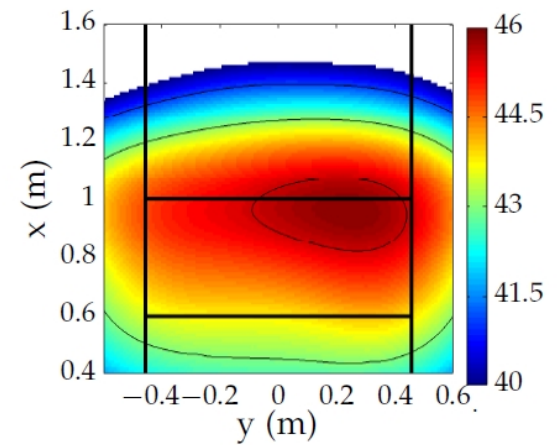
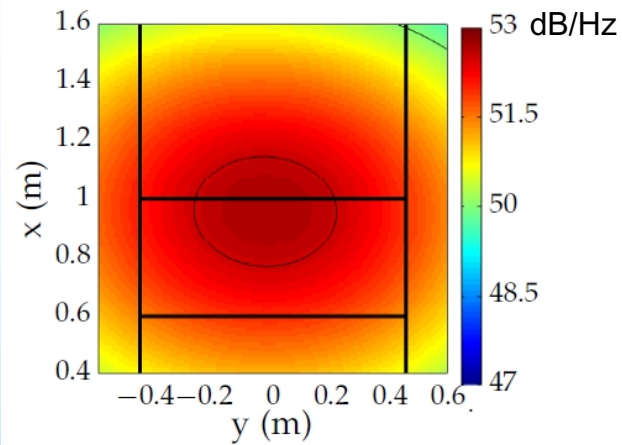
# ***Results from Adelaide / UNSW Analysis Method***

- Techniques:
  1. Conventional Beamforming
  2. CLEAN-SC Deconvolution
- Shear layer correction – method of Amiet (1978)
- Deconvolution – CLEAN-SC (Sijtsma, 2007), beamwidth = 4 cm
- Spatial Integrations:
  - Beamform integrations – method of Brooks and Humphreys (1999)
  - CLEAN-SC integrations – direct summation of source maps



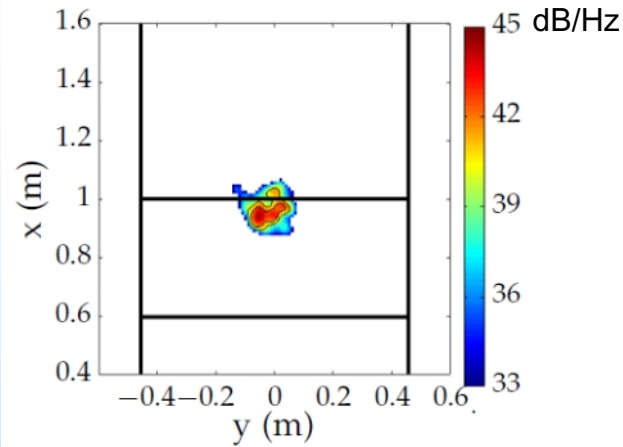
## Results from Adelaide / UNSW

- Conventional Beamforming
- Diagonal Removal
- Background Removal
- 1/3-octave Bands

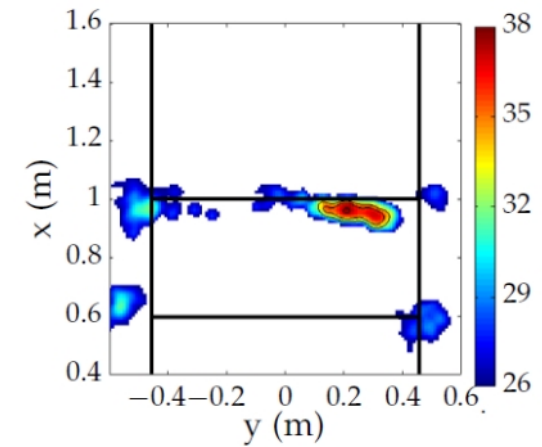


## Results from Adelaide / UNSW

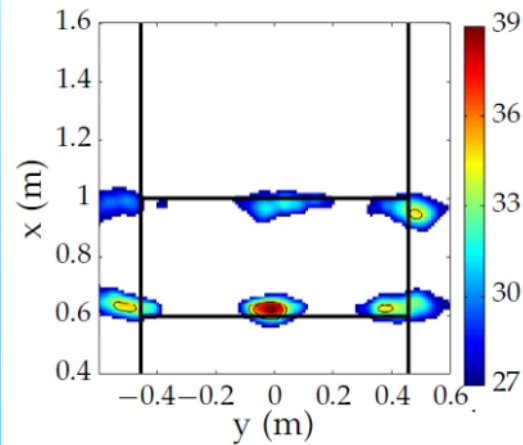
- CLEAN-SC
- Diagonal Removal
- Background Removal
- 1/3-octave Bands



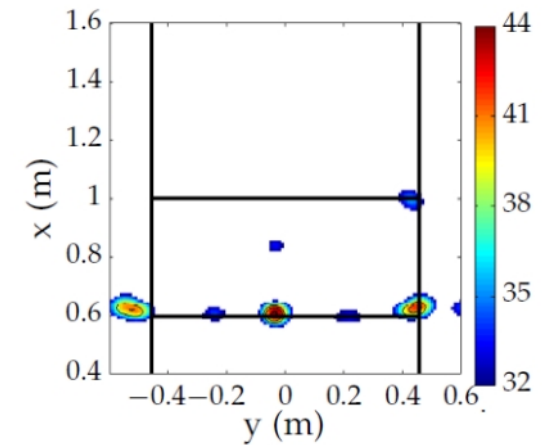
(a) 3150 Hz



(b) 8000 Hz



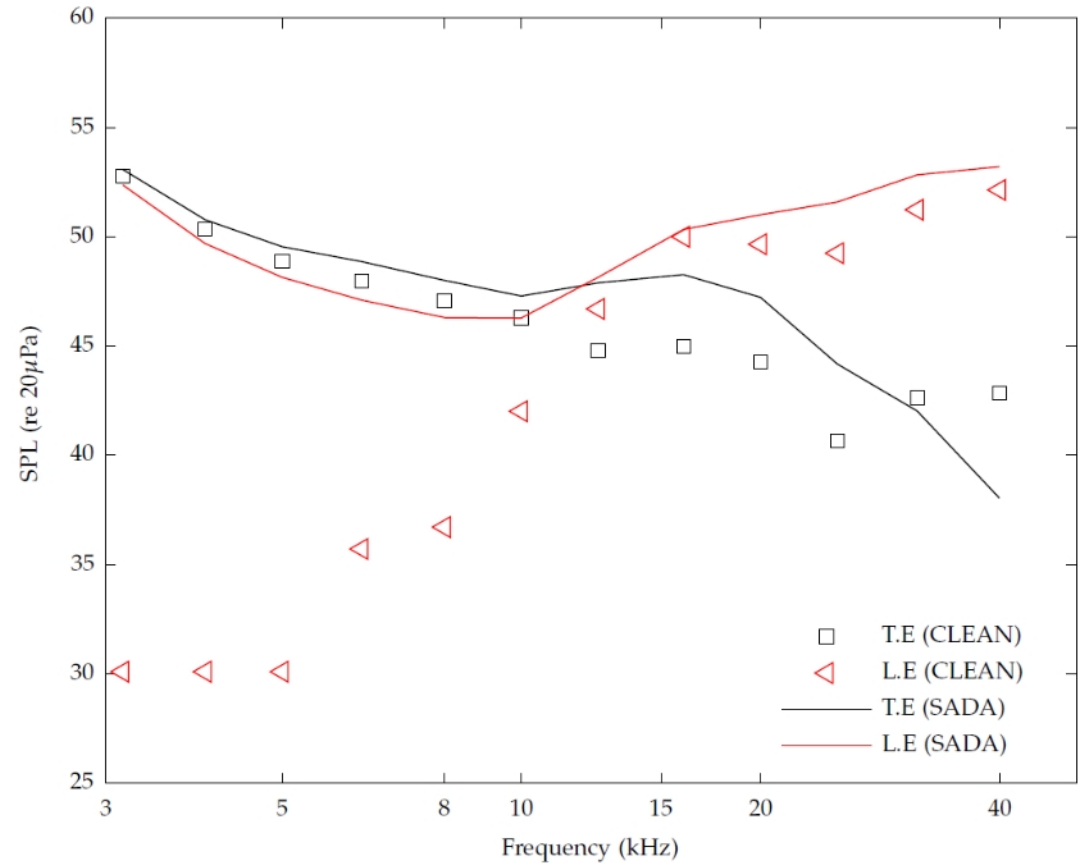
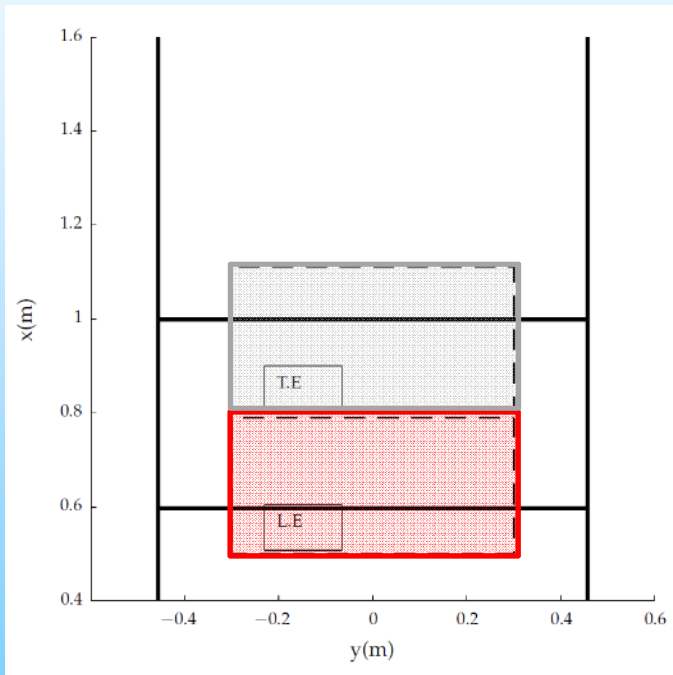
(c) 12500 Hz



(d) 20000 Hz

# Results from Adelaide / UNSW

- Spatial Integrations



**Results from Plateforme d'Antennerie  
AéroAcoustique, Laboratoire Vibrations Acoustique,  
INSA-Lyon, France  
(*Antonio Pereira, Quentin Leclère*)**



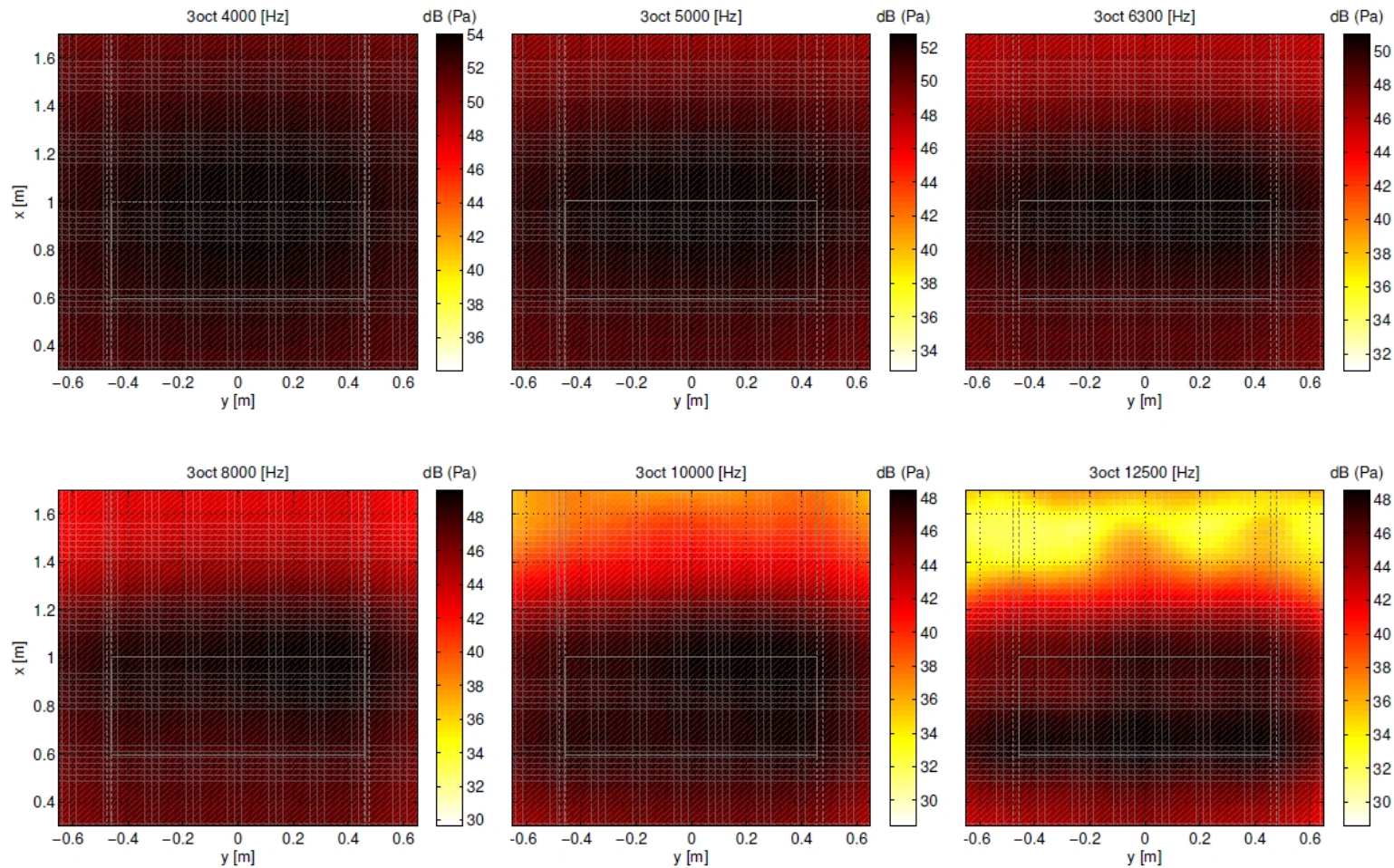
# ***Results from INSA-Lyon Analysis Method***

- Techniques:
  1. Conventional Beamforming
  2. Deconvolution using Non-negative Least Squares (NNLS)
  3. Inverse Method using Sparse Bayesian Reconstruction ( $\ell_p$ -norm with  $p = 0.9, 1.1$ )
- Shear layer correction – method of Amiet (1978)
- Background subtraction – method of Bahr and Horne (2015)
- Spatial Integrations – direct summation of source maps



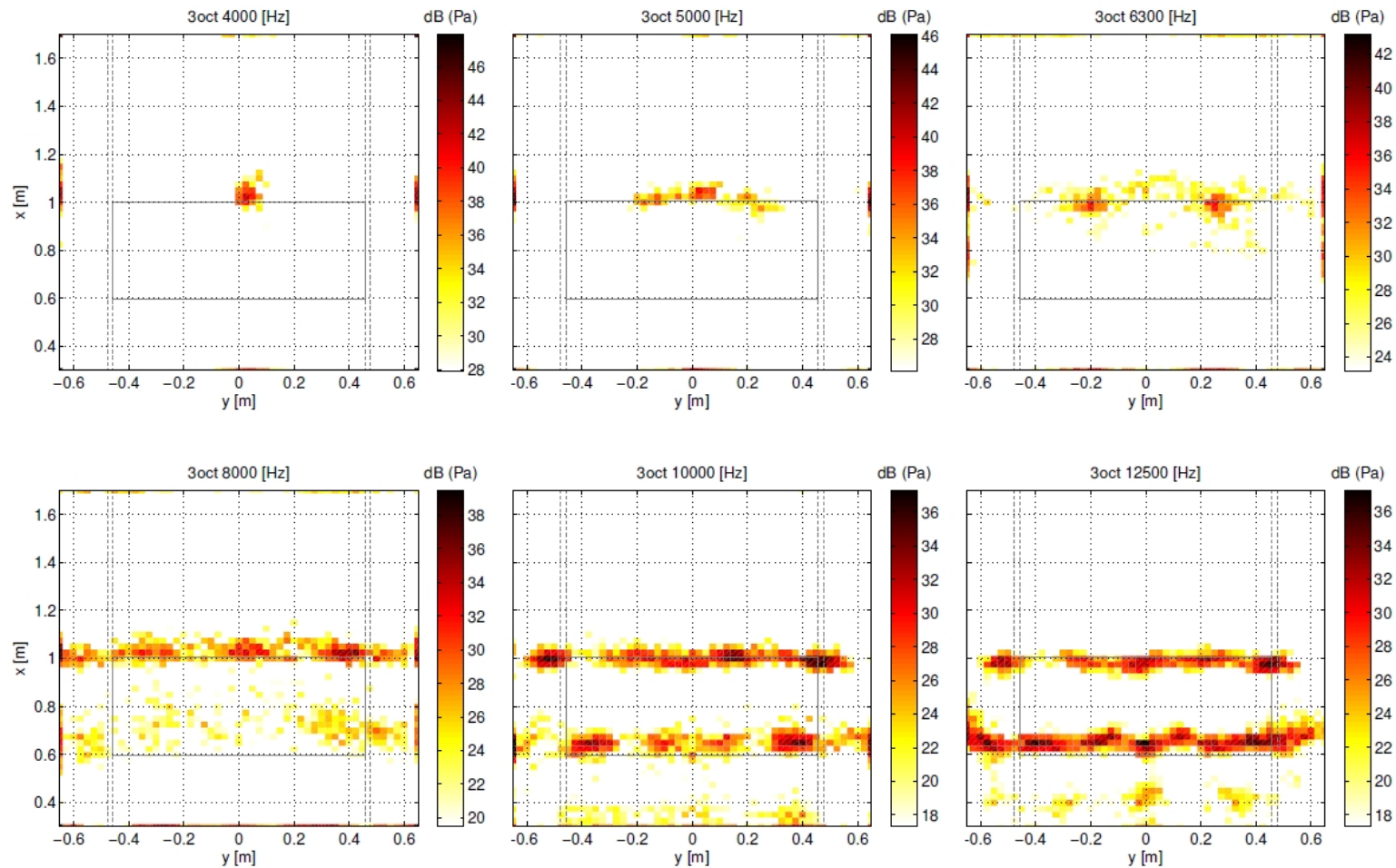
# Results from INSA-Lyon

[4 to 12.5 kHz] - Beamforming (background subtraction)



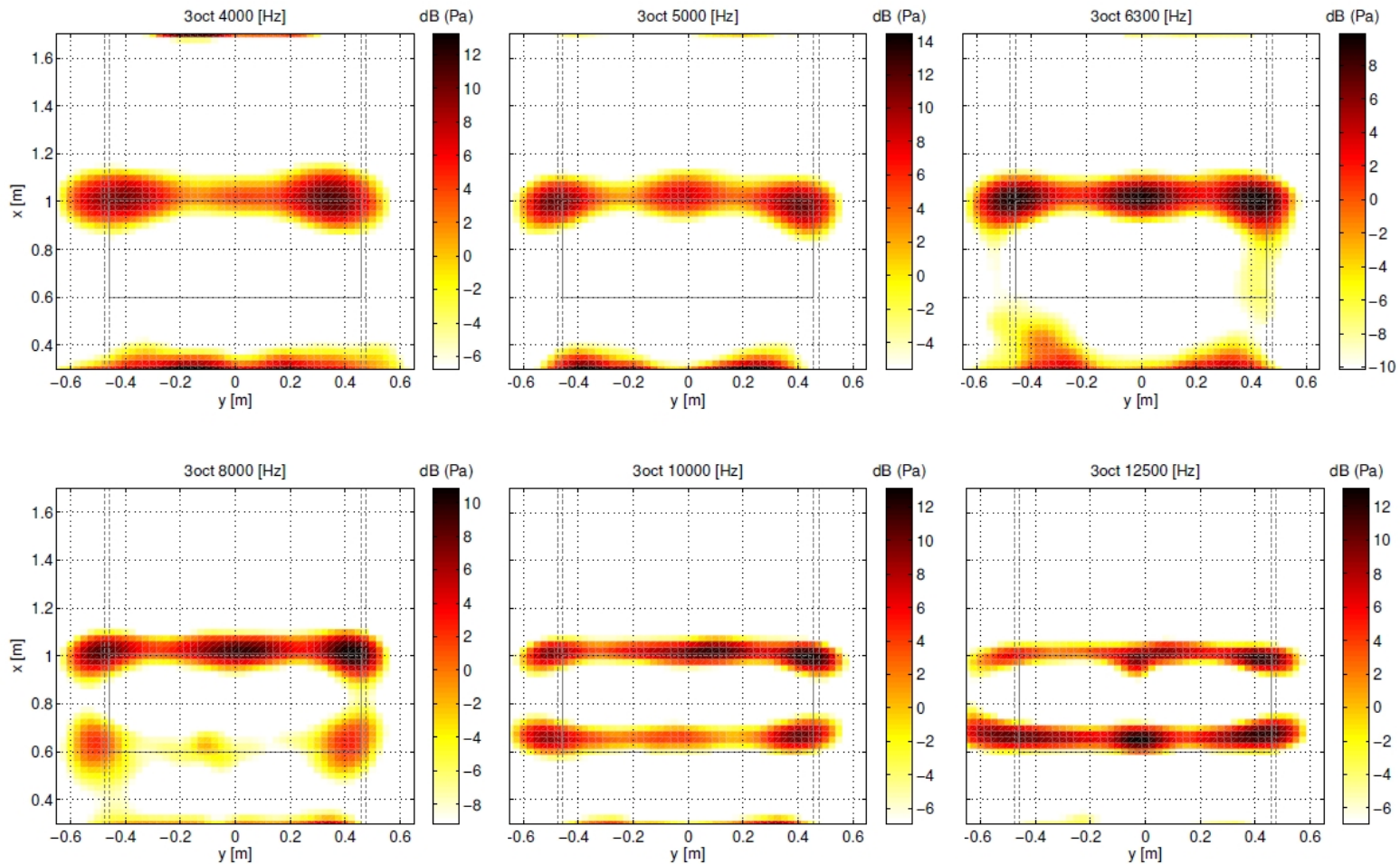
# Results from INSA-Lyon

[4 to 12.5 kHz] - DAMAS nls (background subtraction)



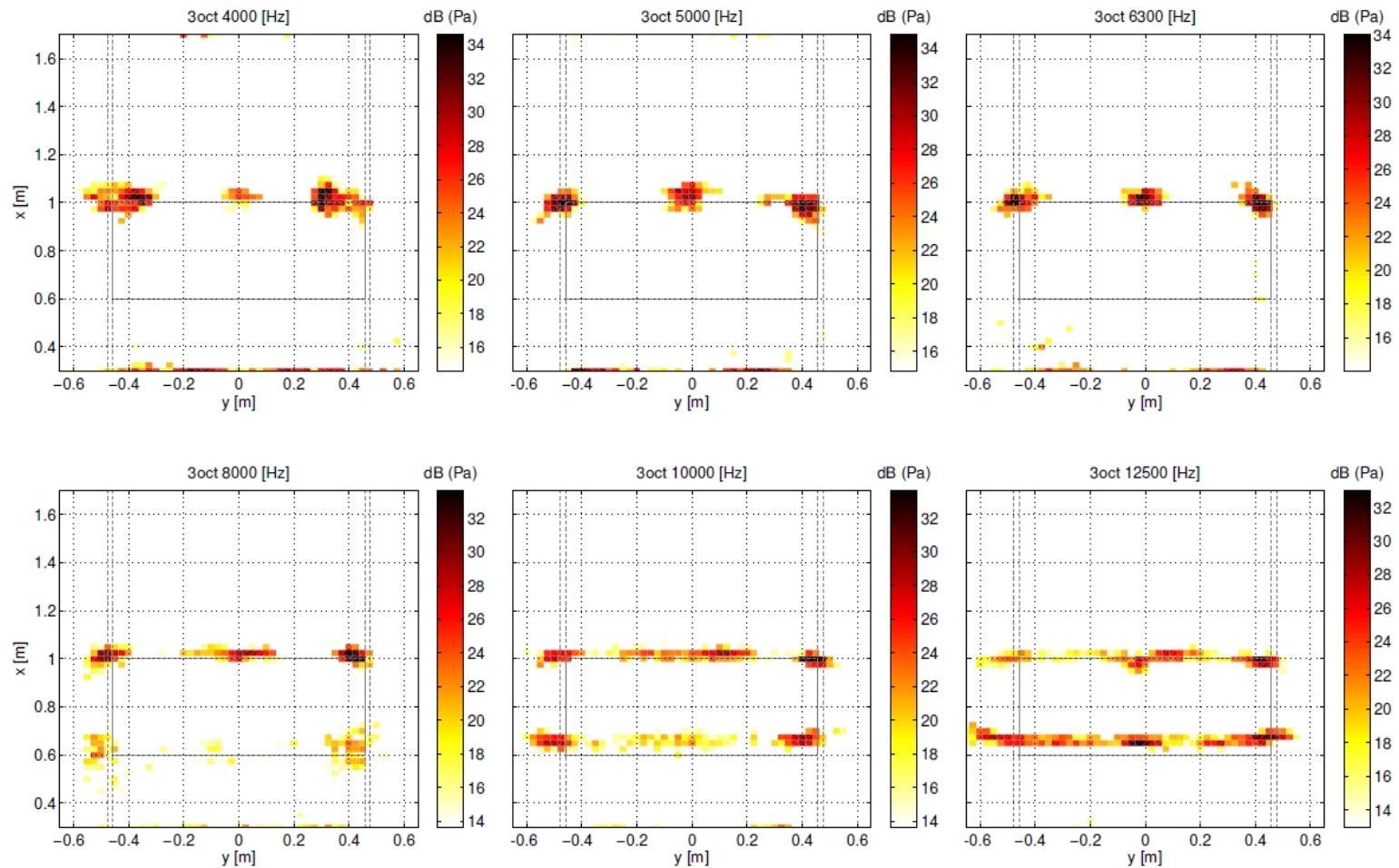
# Results from INSA-Lyon

[4 to 12.5 kHz] -  $\ell_p$  Bayesian ( $p = 1.1$ ) (background subtraction)



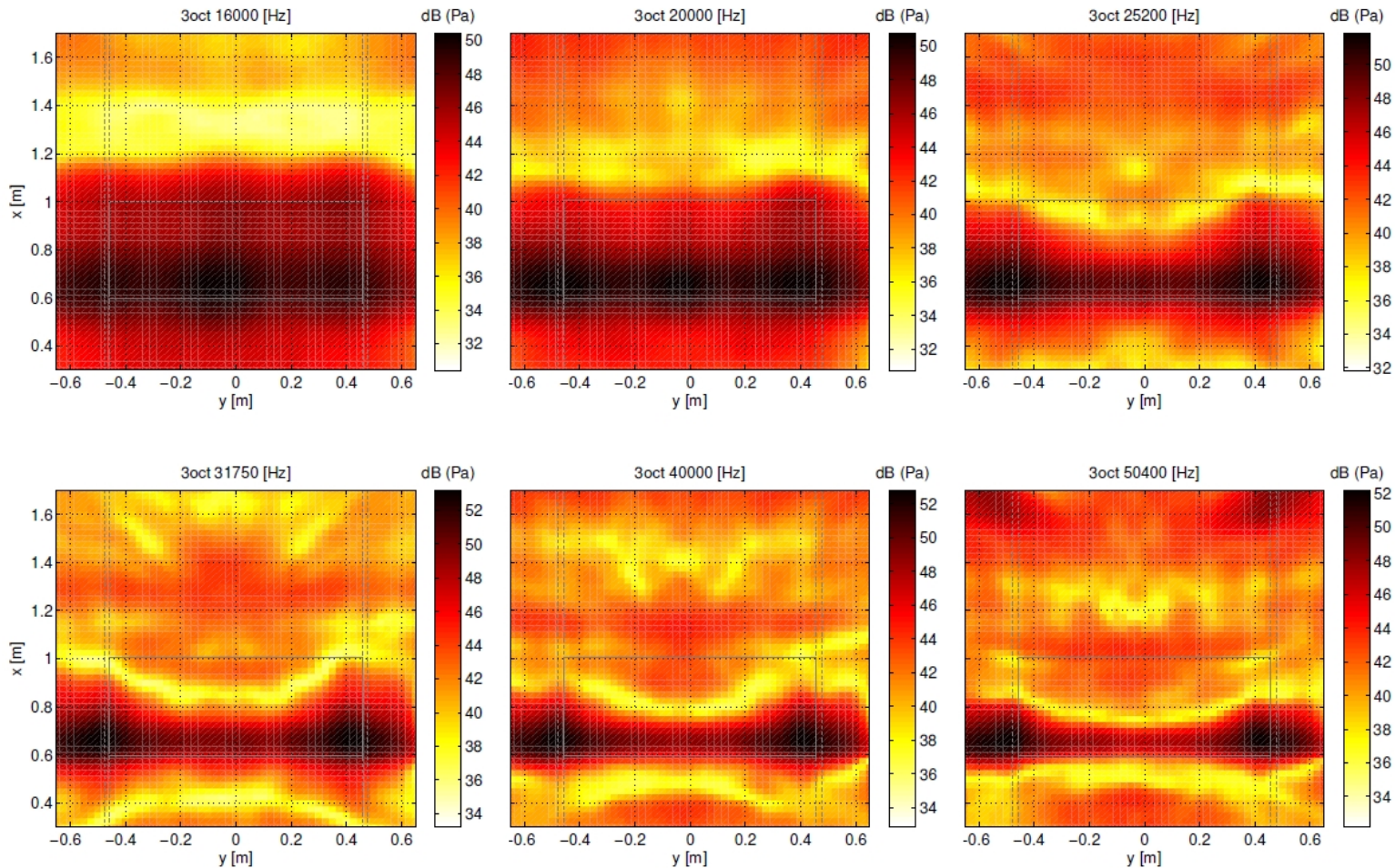
# Results from INSA-Lyon

[4 to 12.5 kHz] -  $\ell_p$  Bayesian ( $p = 0.9$ ) (background subtraction)



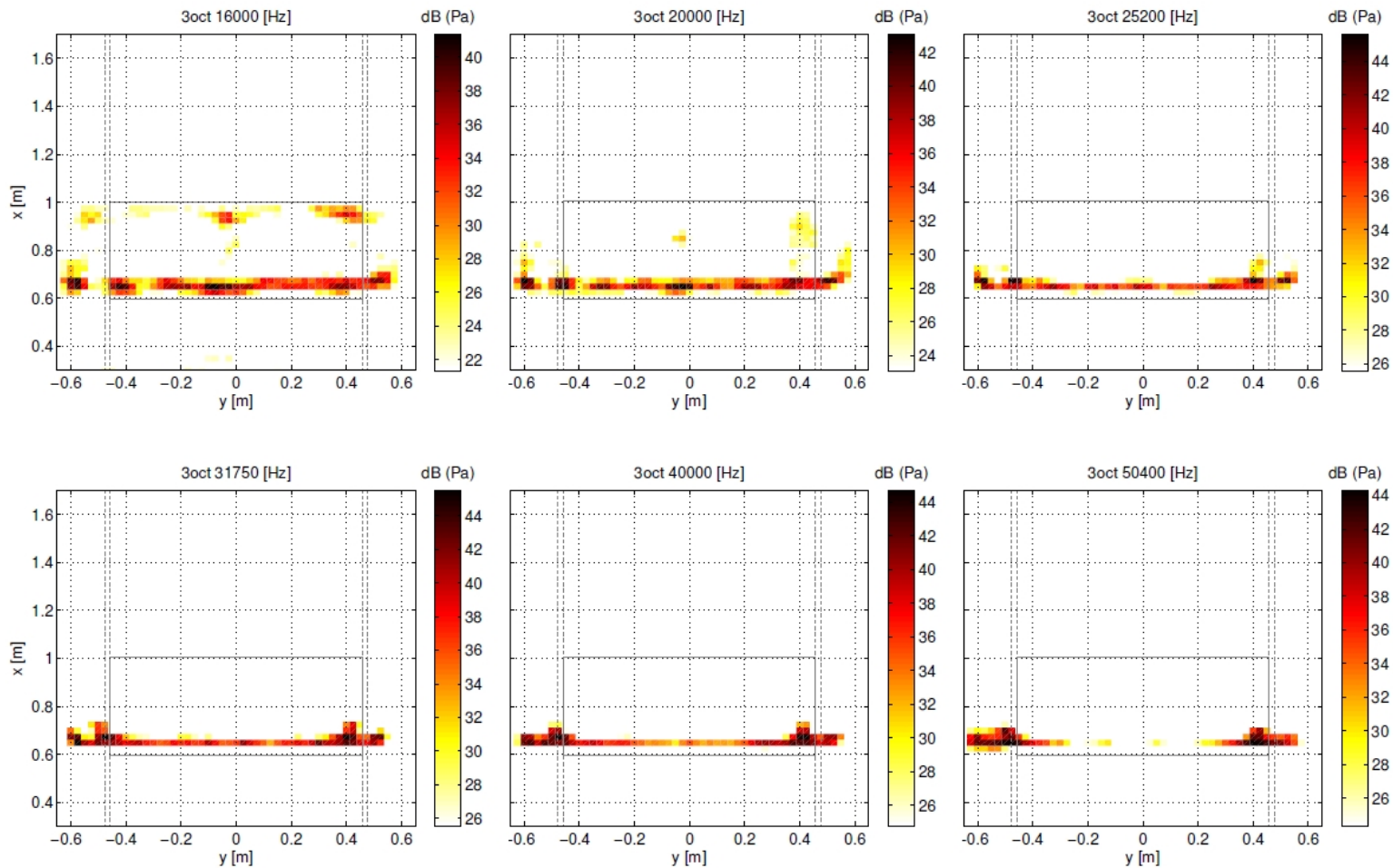
# Results from INSA-Lyon

[16 to 50 kHz] - Beamforming (background subtraction)



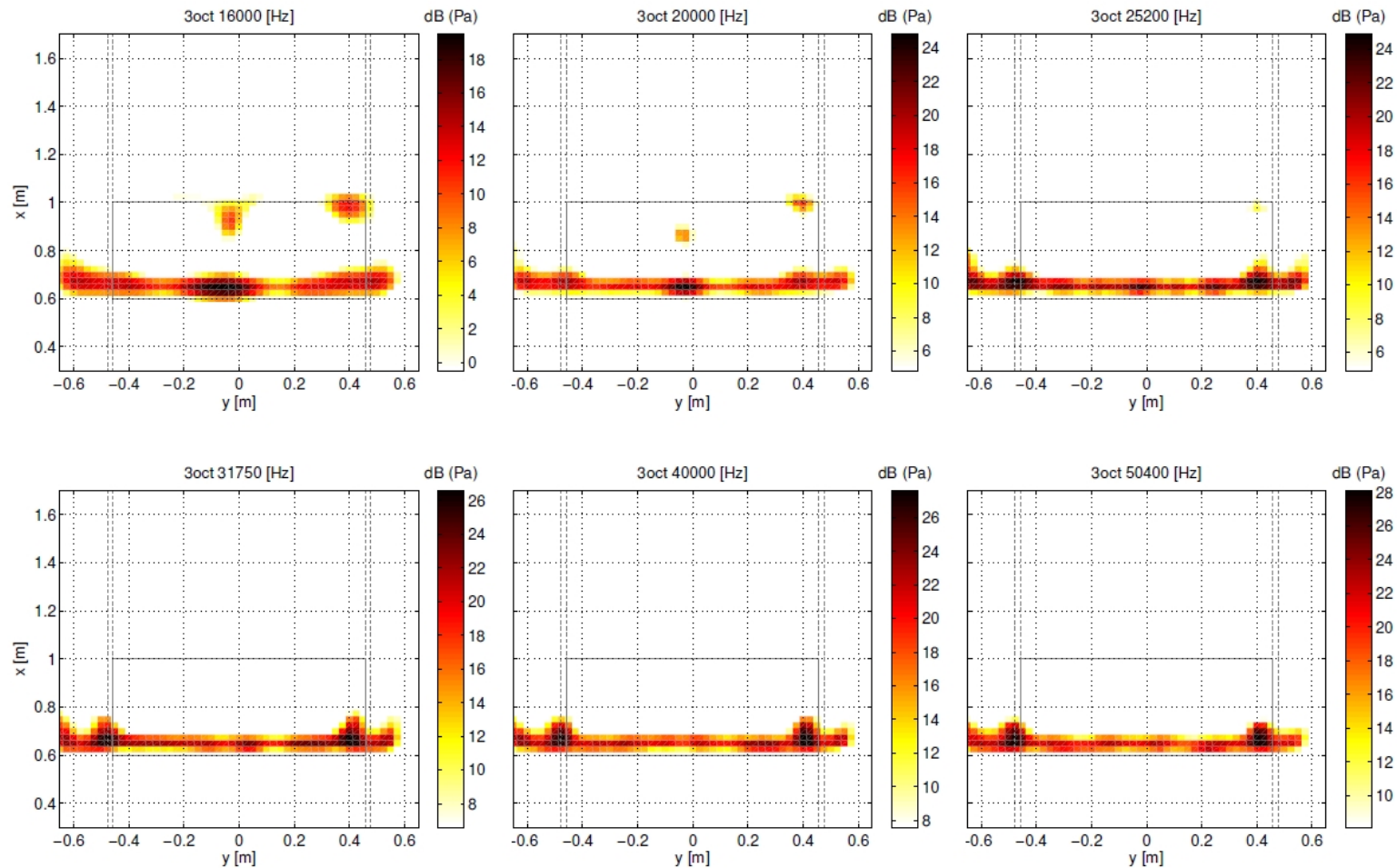
# Results from INSA-Lyon

[16 to 50 kHz] - DAMAS nns (background subtraction)



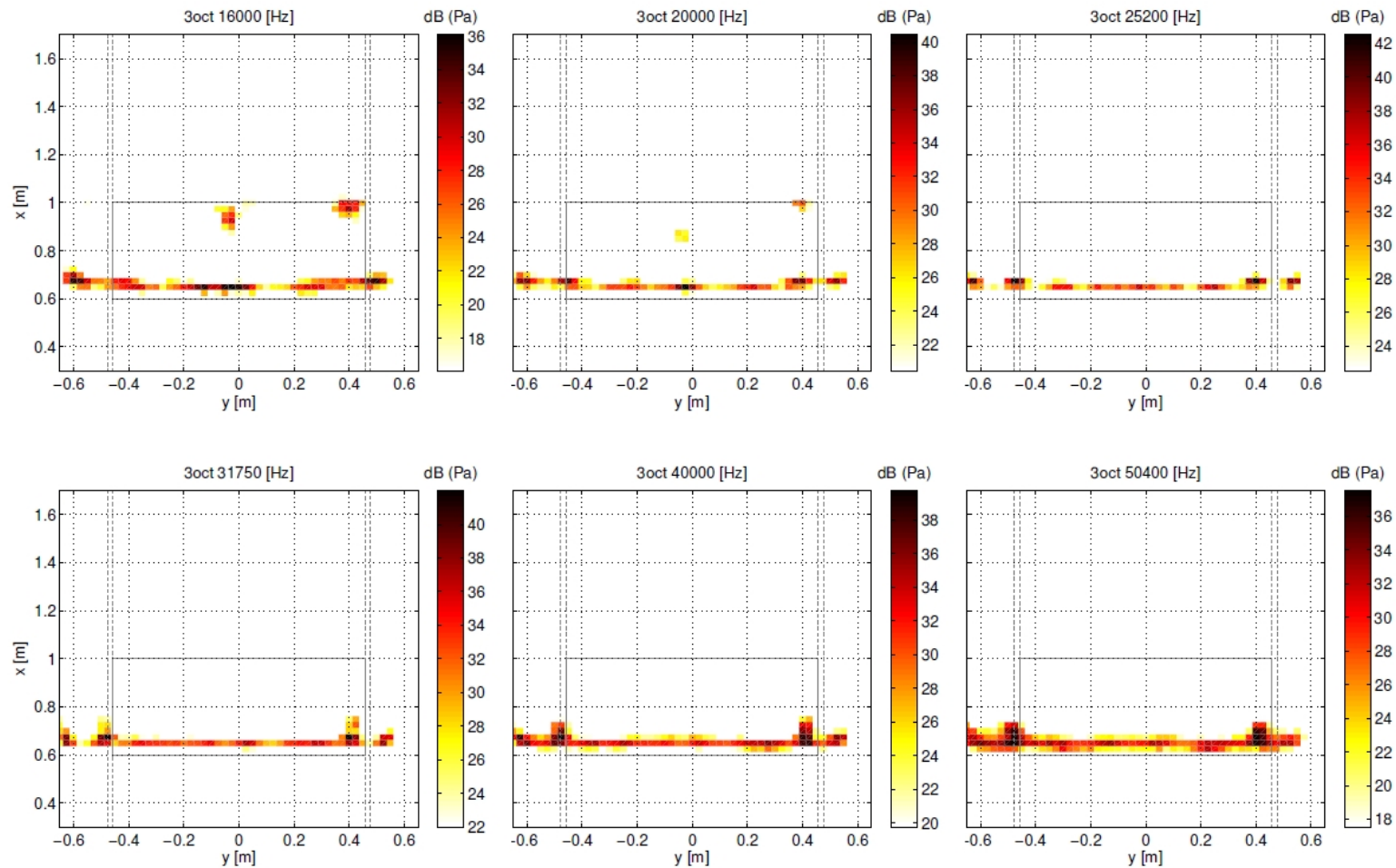
# Results from INSA-Lyon

[16 to 50 kHz] -  $\ell_p$  Bayesian ( $p = 1.1$ ) (background subtraction)



# Results from INSA-Lyon

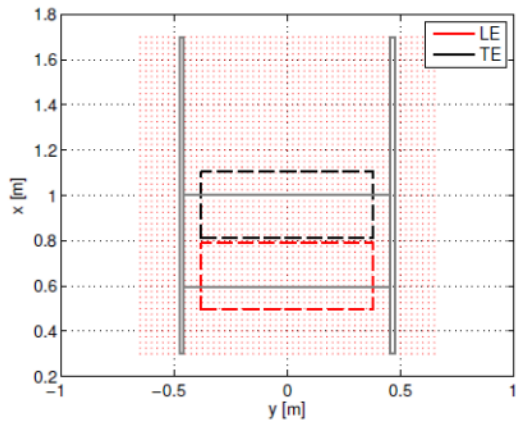
[16 to 50 kHz] -  $\ell_p$  Bayesian ( $p = 0.9$ ) (background subtraction)



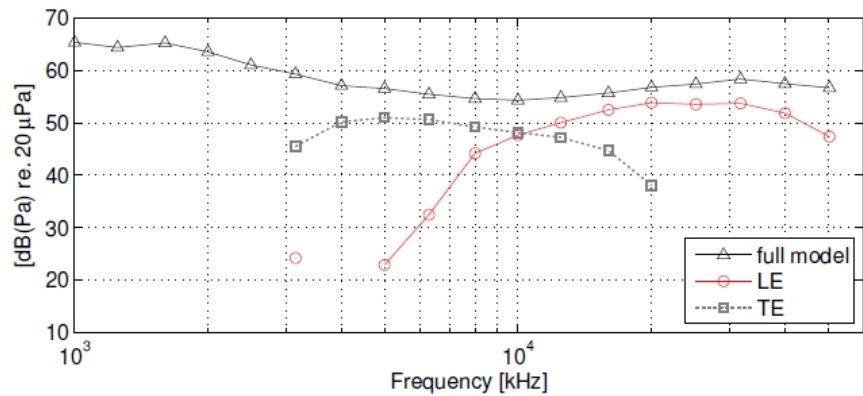


# Results from INSA-Lyon

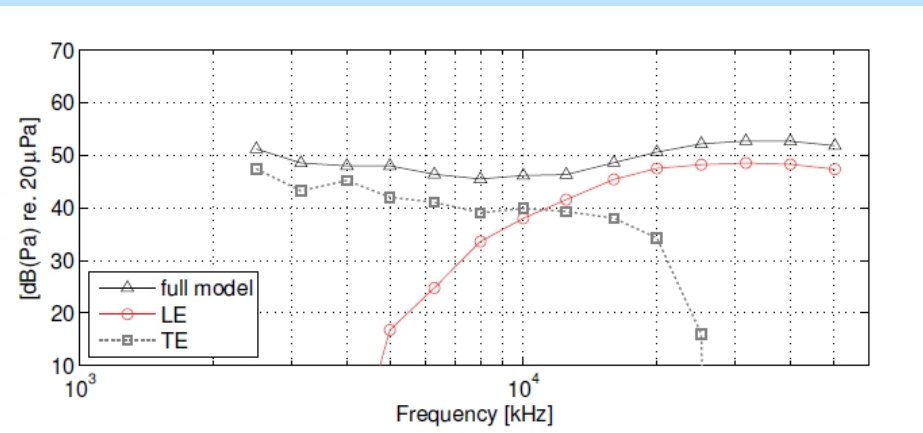
## Source Integrations



Integration Regions

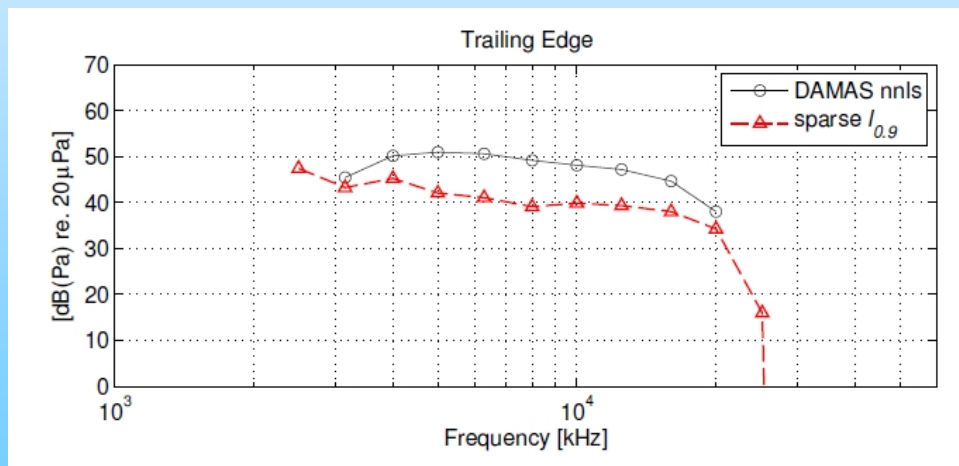
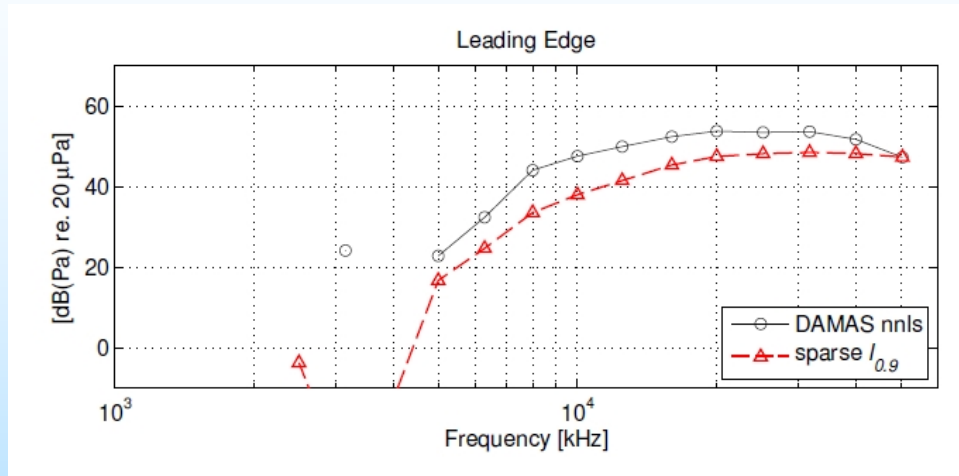


NLS Algorithm



Sparse Bayesian Reconstruction

# Results from INSA-Lyon Source Integrations



**Results from NASA Langley – 2016**  
*(Christopher Bahr)*



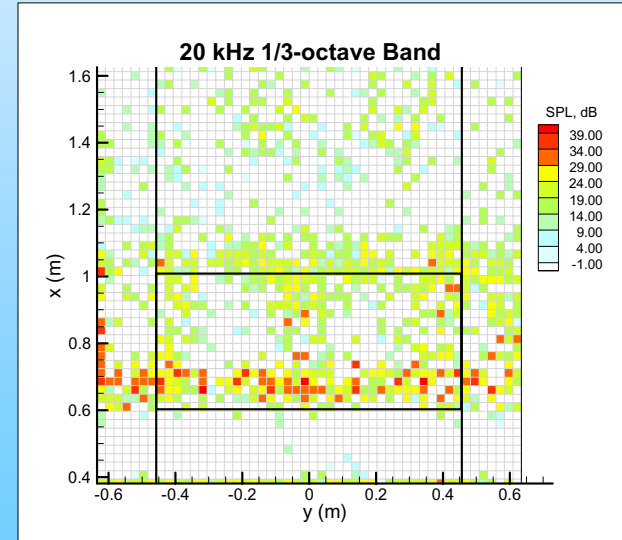
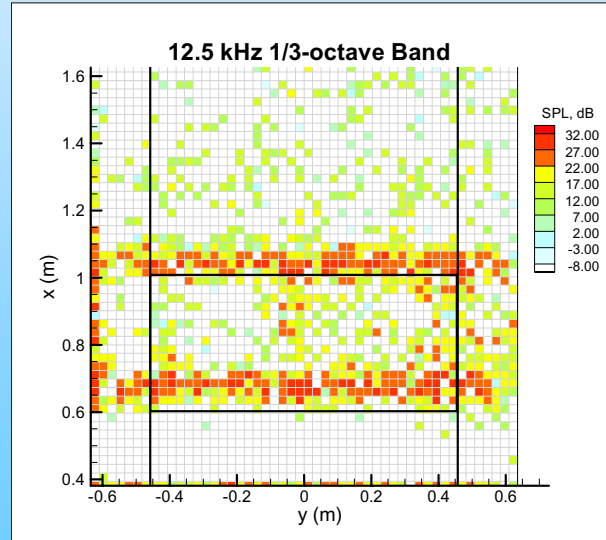
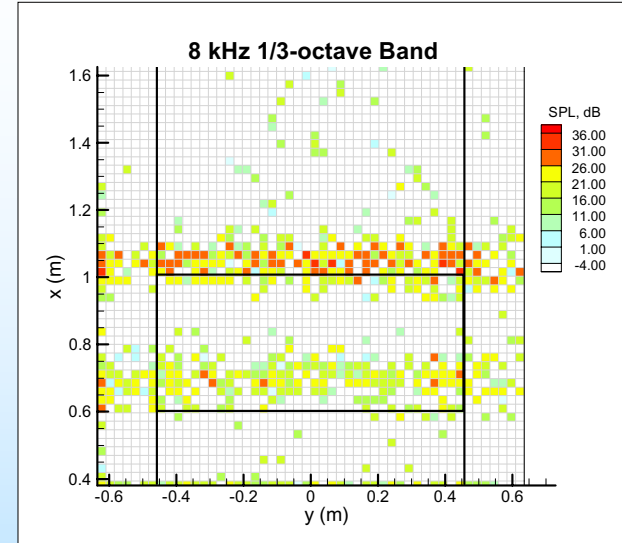
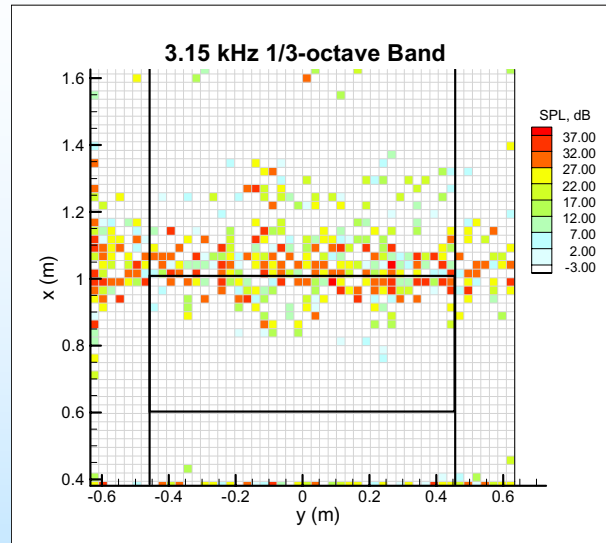
## ***Results from NASA Langley – 2016 Analysis Methods***

Method	DAMAS	Diagonal Removal	Conventional Background Subtraction	Eigenvalue Background Subtraction	Band Combining
1	✓	✓	✓		✓
2	✓		✓		✓
3	✓			✓	✓
4	✓			✓	

- Used dataset-supplied shear layer corrections / steering vectors
- Band Combining – 7 spectral bins summed prior to DAMAS application
- Spatial Integrations - direct summation of source maps

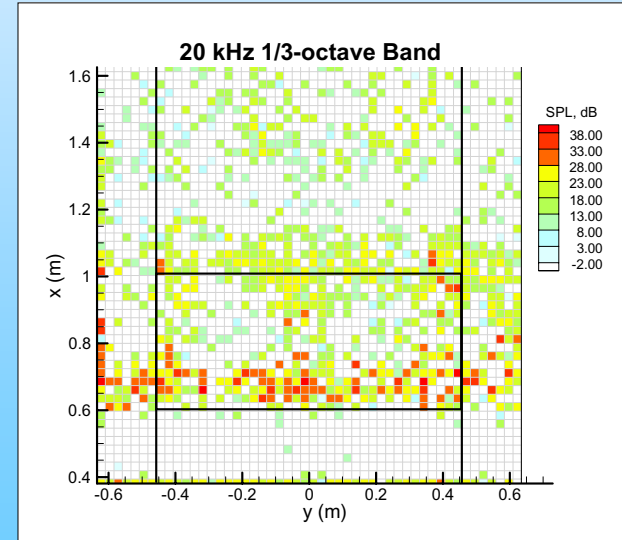
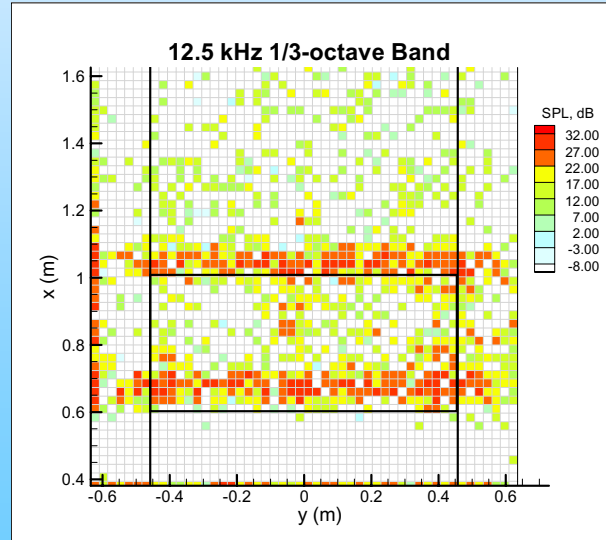
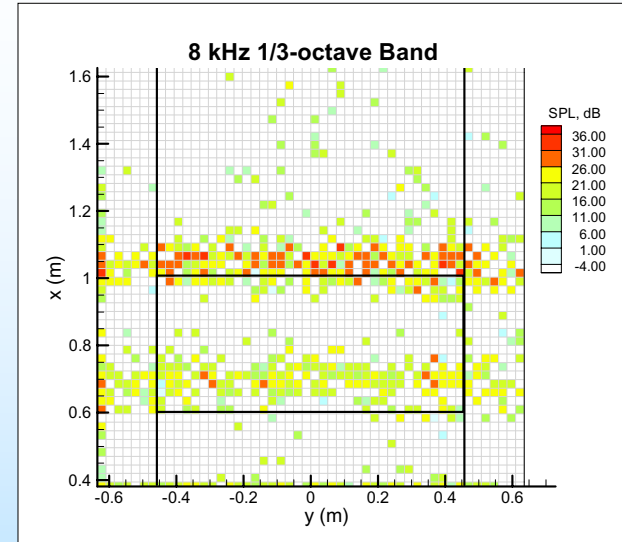
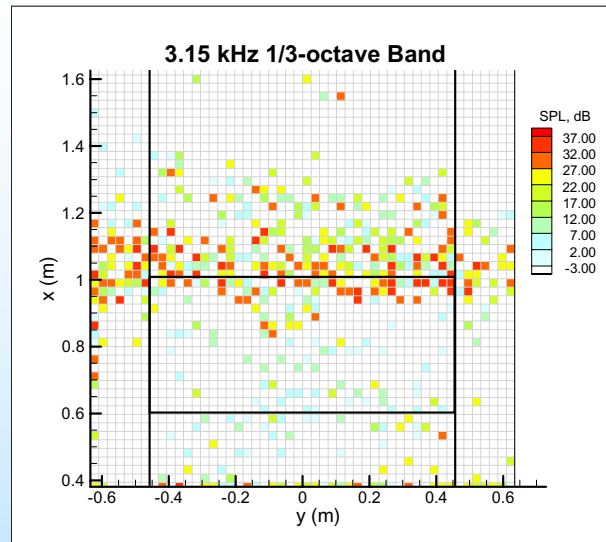
# Results from NASA Langley – 2016

- Method 1
- DAMAS
- Diagonal Removal
- Conventional Background Subtraction
- Band Combining
- 200 iterations



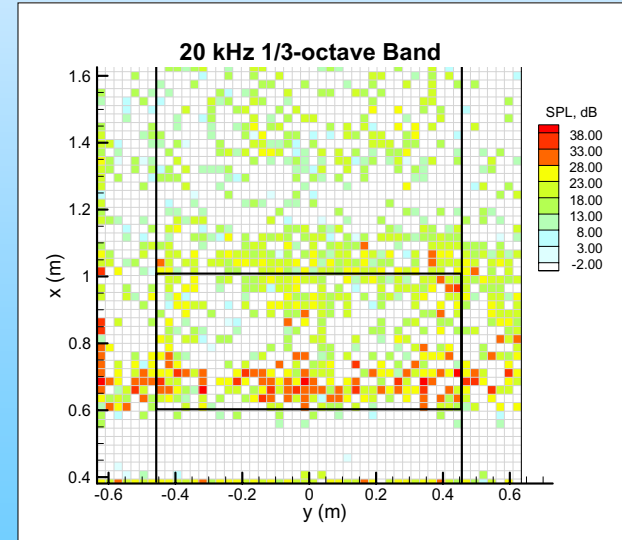
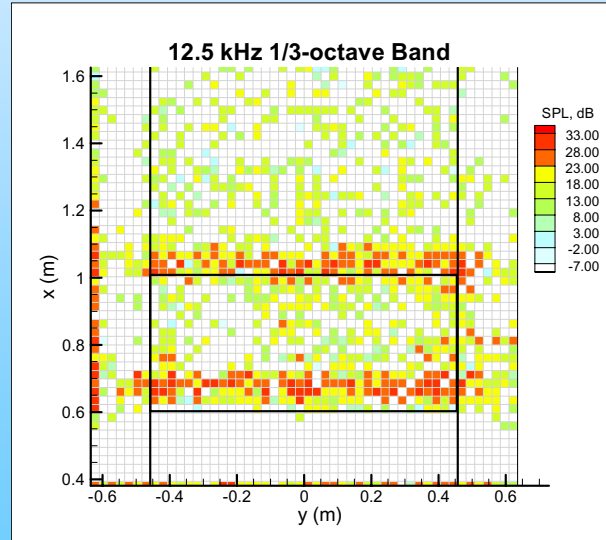
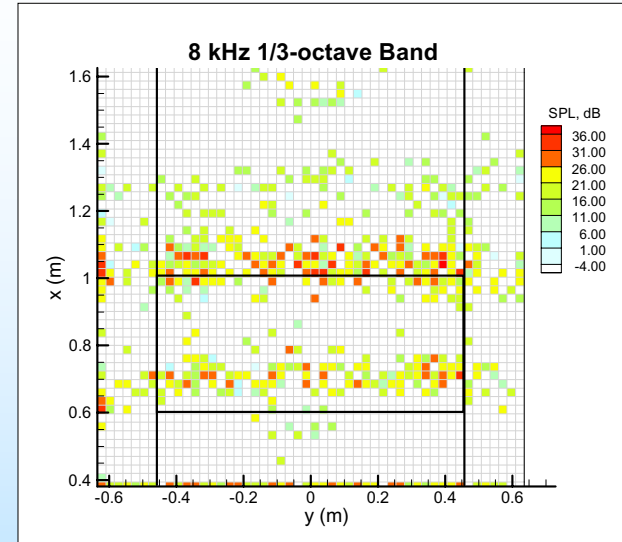
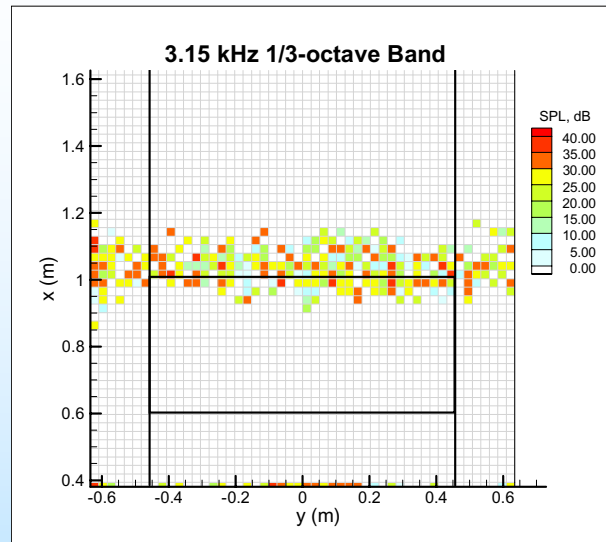
# Results from NASA Langley – 2016

- Method 2
- DAMAS
- No Diagonal Removal
- Conventional Background Subtraction
- Band Combining
- 200 iterations



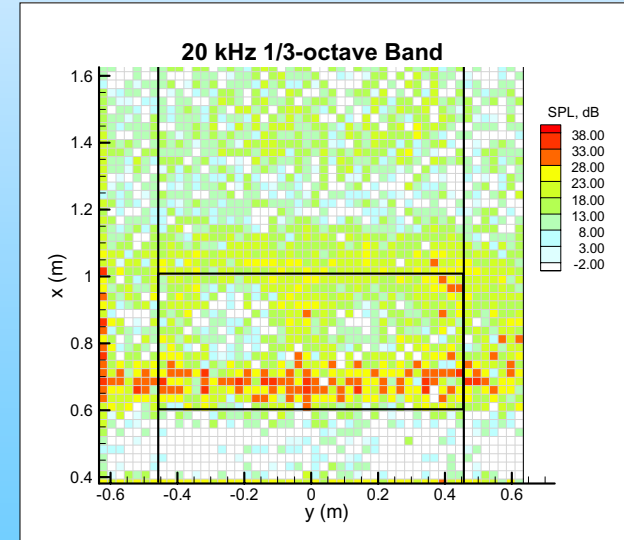
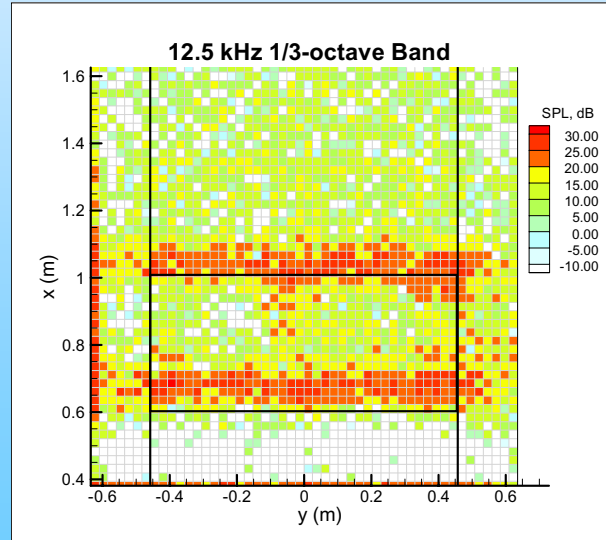
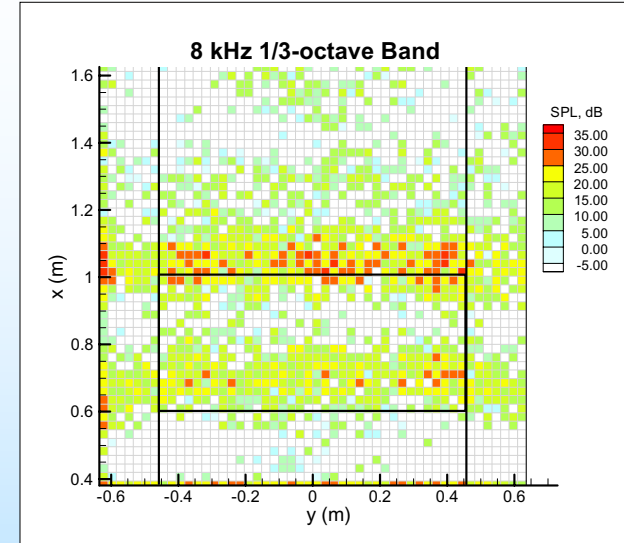
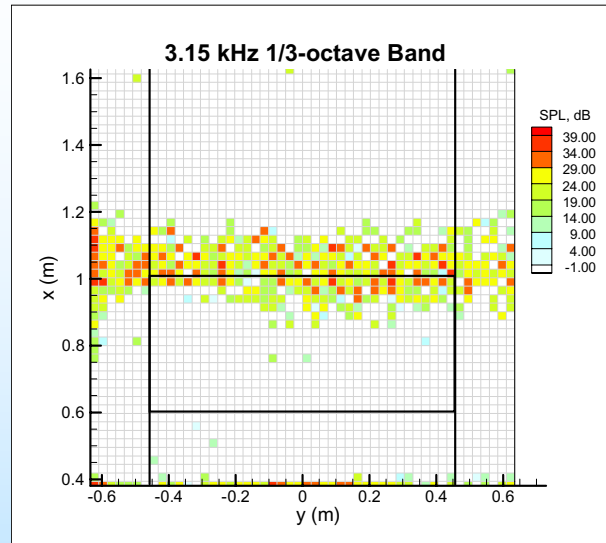
# Results from NASA Langley – 2016

- Method 3
- DAMAS
- No Diagonal Removal
- Eigenvalue Background Subtraction
- Band Combining
- 200 iterations



# Results from NASA Langley – 2016

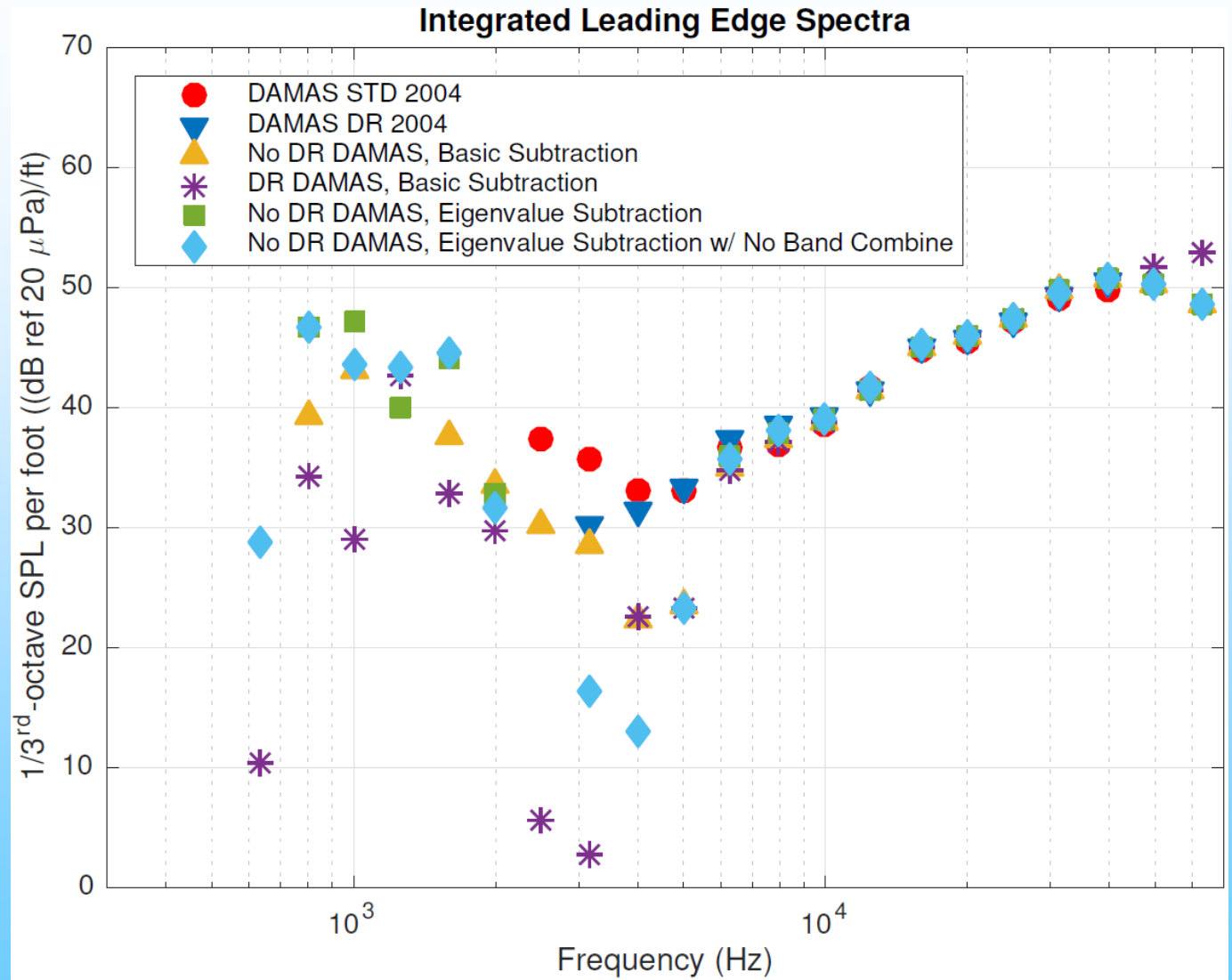
- Method 4
- DAMAS
- No Diagonal Removal
- Eigenvalue Background Subtraction
- No Band Combining
- 200 iterations





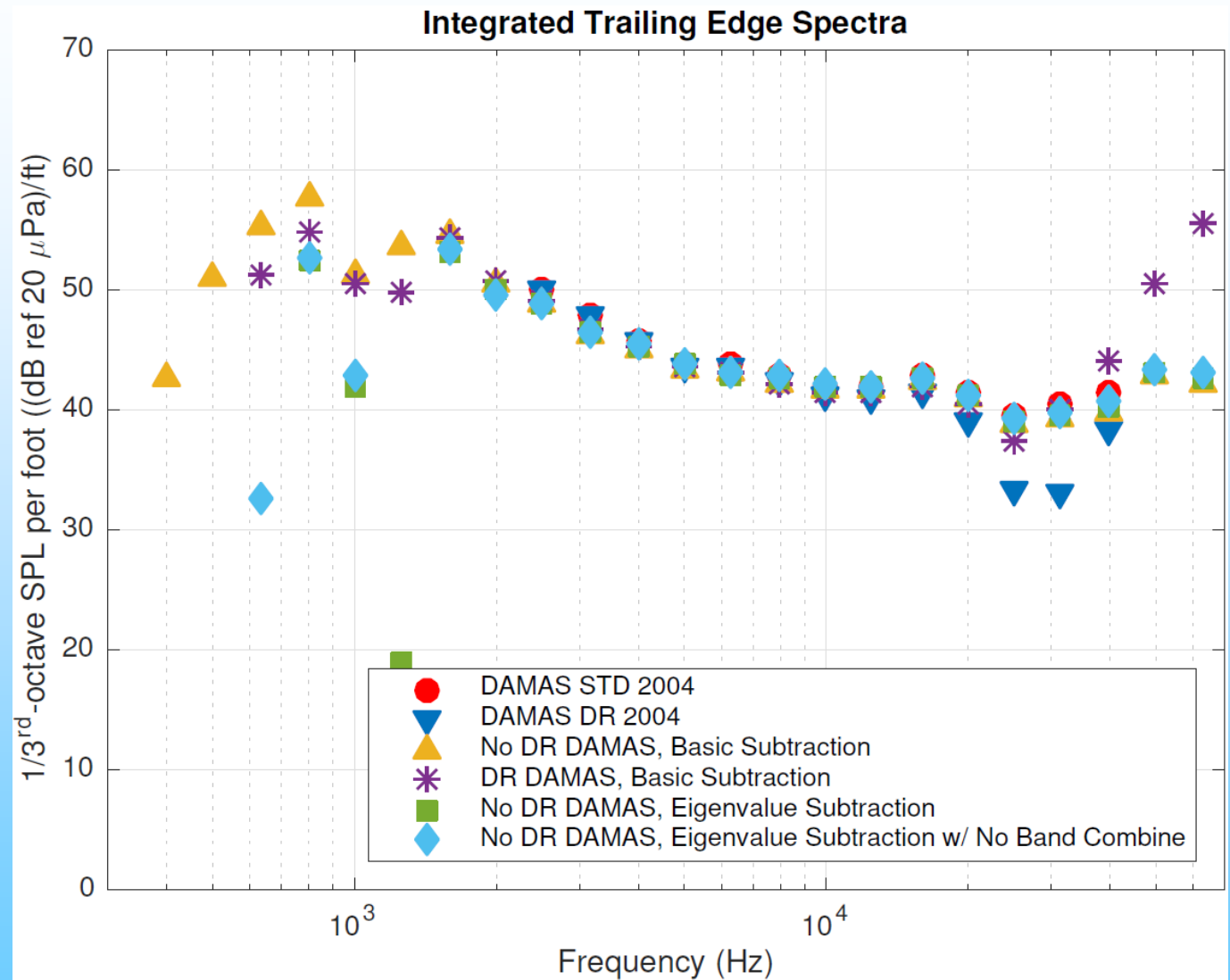
# Results from NASA Langley – 2016

- Leading Edge Spatial Integration



# Results from NASA Langley – 2016

- Trailing Edge Spatial Integration



## ***General Comments / Discussion***

- A variety of analysis methods have now been applied to the benchmark
- In general, many of the visual presentations of source locations are similar between the various analyses
- Most techniques appear to capture consistent source power via spatial integrations at mid- and upper-frequencies
- There are some discrepancies in the integrated source power at lower frequencies, particularly when using DAMAS-types of analyses – causes?
- Traditional CSM diagonal removal can sometimes degrade the visual presentations and lead to discrepancies in integrated source power, especially at higher frequencies

